

a) Anderson 8.16

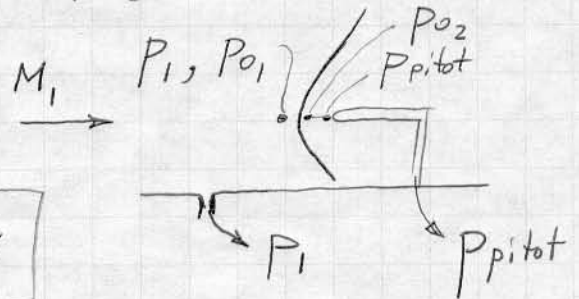
Static $p = 0.1 \text{ atm}$, $P_{\text{pitot}} = 1.13 \text{ atm}$

Since $\frac{P_{\text{pitot}}}{p} = 11.3 > \left[1 + \frac{\gamma-1}{2} M^2\right]^{\frac{\gamma}{\gamma-1}} = 1.89$

the flow is clearly supersonic. $P_{\text{pitot}} = P_{02}$ behind bow shock.

From Anderson App. B,

$\frac{P_{\text{pitot}}}{P_1} = \frac{P_{02}}{P_1} = 11.3$ occurs at $M_1 = 2.9$



b) Anderson 8.17

Across a shock, $h_{01} = h_{02}$

$$h_1 \left[1 + \frac{\gamma-1}{2} M_1^2\right] = h_2 \left[1 + \frac{\gamma-1}{2} M_2^2\right]$$

$$c_p T_1 \left[1 + \frac{\gamma-1}{2} M_1^2\right] = c_p T_2 \left[1 + \frac{\gamma-1}{2} M_2^2\right]$$

From App. B for $M_1 = 36$:

$M_2 = 0.3787$, $\frac{T_2}{T_1} = 252.9$

$$T_{02} = T_2 \left[1 + \frac{\gamma-1}{2} M_2^2\right] = T_1 \cdot \frac{T_2}{T_1} \left[1 + \frac{\gamma-1}{2} M_2^2\right] = 300\text{K} \cdot 252.9 \left[1 + 0.2 \cdot 0.3787^2\right]$$

$T_{02} = 78046 \text{ K}$ toasty

This result is surely invalid, since air will no longer be a perfect gas at this temperature, due to ionization and/or chemical changes. So the shock relations used to get this result are not valid.